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REMARKS FOR RECORD

16 September 1954

SUBJECT: Meeting of Upper Atmosphere Rocket Research Panel

1. At the invitation of Air Force Scientific Advisory Board, I attended a meeting of the Upper Atmosphere Rocket Research Panel on 3 September at the Naval Research Laboratory, Washington. The two areas items of primary concern were: 1) Higher Altitude and Satellite Vehicles, 2) International Geophysical Year, and the logistics therefor.

2. This panel was formed in 1948 by representatives of various organizations which were concerned with upper atmosphere research through rocketry. It is chaired by Dr. James A. Van Allen who is chairman of the Department of Physics, State University of Iowa. The panel has no representation from Aberdeen proving ground, University of Michigan, Naval Ordnance Laboratory, General Electric Company, Naval Research Laboratory, Air Force Cambridge Research Center, California Institute of Technology, and Harvard Observatory. It has no official connection with any government organization or agency but is informally supported by the Office of Naval Research. In connection with the forthcoming International Geophysical Year, it has been assigned the function of developing the United States program for upper atmosphere research through the use of high altitude rockets.

3. At the 3 September meeting, there were in addition to the panel itself, representatives from the Office of Defense, United States Navy, the Ordnance Arsenal, The Rand Corporation, the Office of Naval Research, Aerophysics Development Corporation, Air Force AFSC, NSC, National Science Foundation, etc.

4. Dr. Van Allen opened the discussion of the first agenda item - Higher Altitude and Satellite Vehicles - by a discussion of past upper atmosphere research with rockets. He cited the extensive use of V-2s which could carry 2,000 pounds payload to a 120 miles altitude but the supply of which has now been exhausted. He mentioned the ATICorporal which was not used as a single stage vehicle, primarily because it could carry only 25 pounds to an altitude of 140 miles. The ATICorporal was however, used as a second stage in combination with the V-2 and reached an altitude of 210 miles. Since combustion of the V-2 ceases, the other rocket stages have been developed and used as civilian research instruments. These are the Viking which in its present version can carry 100 pounds payload to 135 miles

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1. At the invitation of Air Force Scientific Advisory Board, I attended a meeting of the Upper Atmosphere Rocket Research Panel on 3 September at the Naval Research Laboratory, Anacostia. The two agenda items of primary concern were:
1) High Altitude and Satellite Vehicles, 2) International Geophysical Year, and the logistics therefor.

2. This panel was formed in 1953 by representatives of various organizations which were concerned with upper atmosphere research through rockets. It is chaired by Mr. James A. Van Allen who is Chairman of the Department of Physics, State University of Iowa. The panel has representation from Aberdeen Proving Grounds, University of Michigan, Naval Research Laboratory, General Electric Germany, Naval Research Laboratory, Air Force Cambridge Research Center, California Institute of Technology, and Harvard Observatory. It has no official connection with my Department of Defense organization but is informally supported by the Office of Naval Research. In connection with the forthcoming International Geophysical Year, it has been assigned the function of developing the United States program for upper atmosphere research through the use of high altitude rockets.

3. At the 3 September meeting, there were in addition to the panel itself, representatives from the Chief of Engineers, United States Army, the Redstone Arsenal, the Rand Corporation, the Office of Naval Research, Aerophysics Development Corporation, Air Force AFSC, NASA, National Science Foundation, etc.

4. Dr. Van Allen opened the discussion of the first agenda item - High Altitude and Satellite Vehicles - by a discussion of past upper atmosphere research with rockets. He cited the extensive use of the R-2 which could carry 2,000 pounds payload to a 150 mile altitude but the supply of which has now been exhausted. He mentioned the WAC Corporal which was not used as a single stage vehicle, primarily because it could carry only 25 pounds to an altitude of 140 miles. The WAC Corporal was however, used as a second stage in combination with the V-2 and achieved an altitude of 240 miles. Since exhaustion of the V-2 supply, the other rockets have been developed and used as two stage vehicles. These are the Viking which in its present form can carry 100 pounds payload to 135 miles.

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altitude and in its forthcoming; notification will carry 500 pounds to 125 miles altitude. The second rocket is the Aerobee which can carry 100 pounds payload to 65 miles altitude. In addition to these two, there is also the "Black" which is a system of launching a Jason rocket from a "piggyback" balloon. The launch is made at a balloon altitude of approximately 100,000 feet and the rocket carries 30 pounds of instrumentation to an altitude of 60 miles. Mr. Van Allen concluded his presentation by stating that, from here on, civilian upper atmosphere rocket research will probably be dormant, due to lack of civilian funds, upon the military rockets being developed now by the Department of Defense. He then introduced Mr. Fred Whipple of the Harvard Observatory for a discussion of Earth Satellite Vehicle (ESV).

5. Mr. Whipple stated that the interests of civilian research and of the Department of Defense in 1957 are synonymous. He pointed out the main scientific mission which could be made through the ESV which have application both to civilian science and to defense. These include meteorology, astronomy, solar-terrestrial relations, far ultra-violet and X-ray research, etc. He described the ESV sequence as being in three phases as follows:

(1) The placing in orbit of an observable object (uninstrumented) which could be seen either optically or by radio.

(2) The placing in orbit of an instrumented vehicle, "an unarmed physical laboratory". This phase will be a progressive development starting with a very small vehicle similar to the timer, which will carry telemetering equipment. Larger capacity will be achieved as bigger and more reliable power plants are developed. These latter will probably be nuclear or solar energy plants. Instrumentation starting with simple telemetering will progress through more complicated stages until television and finally a telescope is included. The last stage in this phase will be a remote controlled vehicle.

(3) The final phase of ESV will be the putting in orbit of a manned satellite vehicle.

6. Mr. Whipple emphasized that the placing of a slug in orbit was in fact the first step in the process. He stated that such a vehicle even without instrumentation could produce useful scientific results such as air density data and relative positions on the earth. The main problem in connection with a slug (where there is no place of observation from the earth, which will require much study). This problem will be simplified if the ESV is on either an equatorial or a polar orbit. In regard to

the unmanned instrumented vehicle (Phase 2), he said that the main problems will be development of small reliable power plants; television, the technical design of which is already well along; orientation of the vehicle itself and of the instrumentation carriage; and constant reduction of weight of the equipment to be carried. He stated that the altitude at which the vehicle should orbit will depend upon the purpose envisaged. Theoretically, an altitude of 1,000 miles at a speed of 5 miles per second would be ideal. This would provide a 2-hour orbit.

7. Mr. George Danner of the Mr. Branch, CDR, made the next presentation, concerning high altitude vehicle projects with which CDR is concerned. CDR has three main projects in this field, the first of which is the development of a manned conventional aircraft to operate at a medium altitude of 300 miles. Two designs are presently under consideration - the Douglas 553 and the "curved 155". The former is designed to have an altitude of 700,000 feet. (Note: this field of study is covered in Report N-CDR-1246(30), "Low Altitude and High Speed Study" by Douglas Aircraft Corporation - CDR, 1957(L)). The second CDR project is development of a manned high altitude balloon to operate at 100 to 200,000 feet. The system would be based on the "Skyhook" polyethylene balloon carrying a gondola equipped to sustain one or two men. The third CDR project is for manned space flight and the study of this is being conducted at the Aero Jet laboratory. The first phase of this latter project is called Project 100, the purpose of which is to place an AV in orbit at an altitude of 200 miles in order to measure micrometeorological and atmospheric data at that level as the first step toward higher altitude work. Project 100 has been tentatively approved by the Navy and CDR is going ahead with it in cooperation with the Army. It is now also being coordinated with the Air Force at a very high level. The project calls for the use of the Army Redstone missile (see para. 8 below) as the first stage with the Loki cluster (see para. 9 below) providing the second and third stages. Under Aero Jet leadership, four subsidiary studies are planned to be undertaken as follows:

(1) A Visibility Study to determine the size and weight of the vehicle required at an altitude of 200 miles. It is hoped that Mr. Fred Whipple will be the leader of this.

(2) An Orbital Study to determine the power required, the guidance system, etc. It is hoped that Mr. S. Fred Singer of the University of Maryland will lead this.

(3) A Trajectory-Trajectory Study to determine the final design and the staging requirements.

(b) A launching study to determine where and how the vehicle should be launched, the logistics requirements, and the range risks involved.

Following the completion of these four studies, construction of the actual vehicle will be commenced. It is considered that successful completion of Project [REDACTED] will lead into the launching of a vehicle similar to the latter [REDACTED], i.e., an instrumented vehicle using a polar orbit at an altitude of 200 miles. It is expected that [REDACTED] would remain aloft for ten days while [REDACTED] could probably maintain its orbit for about one month. It is further emphasized that if adequate backing is available for Project [REDACTED] [REDACTED] might well be used during the International Geophysical Year. It is laid great stress on the necessity for the United States being the first in launching an I.G.Y. and said that Project [REDACTED] was absolutely essential to achieve this end.

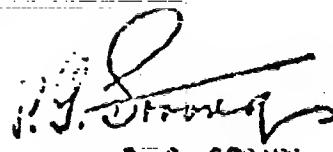
8. Mr. Miller of Redstone Arsenal then gave a briefing on the Redstone missile, the R-17 model currently, as a tactical weapon, has a range of 170 miles with a trajectory apex of 60 miles. If used as a research tool, it has a vertical range of 170 miles with a weight of about 1,50 pounds. The modified [REDACTED] which is now under development will have a vertical range of 270 miles with an instrument weight allowance of 500 pounds. Project [REDACTED] is now considering the use of these missiles as research instruments. It will cost \$20,000 to \$30,000 per missile if more than five projectiles are produced. They have a speed of approximately .600 Mach.

9. The design of the American Development Corporation described by Mr. Miller and his associates [REDACTED] are developing as an air probe project. The vehicle consists of two clusters of solid rockets; the first cluster of seven rockets as the first stage booster and second cluster of four constituting the second stage. It has a speed of Mach 10 and an overload of 10 pounds. Air-launched vertically from an altitude of 10,000 feet, it is designed to continue an altitude of 100 miles. It is planned to use either a flashing light system or a folding radar reflector for tracking purposes. Unstabilized, it is expected to get an altitude of 20 miles. With solid rocket boosters, the second stage develops 1,600 lbs. The missile has yet to be flown vertically as a research instrument. Mr. Miller indicated that the second stage cluster could be adapted to carry 12 pounds of subjected to monitoring equipment.

10. Mr. Miller of [REDACTED] commented briefly on a design which they are currently working on which consists of the [REDACTED] booster as the first stage and a solid rocket as the second stage. It is estimated that this vehicle could attain an altitude of 100,000 feet. It would be very low cost, in the range of \$7,000 to \$10,000.

11. Dr. Holloman of the National Defense Foundation gave a brief account of the situation of the funds for upper atmospheric rocket research. \$4.0 million, three-million thousand dollars is presently available and about 75% of this will be transferred immediately to GPO for procurement and accounting purposes. The balance of the rocket project funds - \$400,000 - will become Available next year.

12. Following adjournment of the panel meeting, I spent a couple of hours with Mr. Van Allen, who had served under me for a time during World War II. Van Allen was one of the key figures in the development of the N-5 and upper-Admiral Parsons and was one of the officers assigned to introduce the fuse to the Pacific Fleet. In our conversation, I mentioned the difficulty that apparently would be encountered in the J-3 program, of assigning scientific objectives which would stand the high G's encountered in flight with solid-fueled rockets, particularly with solid propellant. I mentioned specifically the 100 G's experienced in stage two of the long cluster vehicles. Dr. Van Allen said that there should be no such difficulties. He pointed out that the V-1 fuse contained five miniature radio tubes, a battery and a transmitter, and that this instrument, with approximately no failures, withstanded 20,000 G's when fired from the naval gun mountings. Moreover, he pointed out also that in the I-100 I-100 fusion has been in use for some time, the ocean wave at eastern 30 pounds of instrumentation for oceanographic purposes having been tested with loads up to 100 G's. He said that he saw no difficulty whatever in designing instrumentation for high altitude rockets which could withstand 1,000 G's. In regard to the J-3 program, namely, he agreed thoroughly with Dr. G. C. Seaver, that the alternative first step was to launch a small satellite. He felt that if test and development priority could be established, with guidance and control by an outside high level scientific group, it would almost certainly be possible to put a satellite in orbit by the time of the J-3 and possibly even put up an instrumented vehicle.



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